

HUMMINGBIRD HOVERING ENERGETICS DURING MOULT OF PRIMARY FLIGHT FEATHERS

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Summary

How does a hovering hummingbird compensate for the loss of flight feathers during moult when the mechanism of lift force generation by flapping wings is impaired? The flight performance of five individual ruby-throated hummingbirds with moulting primary flight feathers and reduced wing area was compared with that before their moult. Hummingbirds were flown in reduced air densities using normoxic heliox so that a range of flight energetics was displayed. The rate of moulting and the extent of wing area loss varied among individuals. One female could tolerate a 30% loss of wing area in moulting and flew with only three outer primaries per wing. Further exploratory study using the artificial reduction of wing area, either by cutting the tips of the outer primaries of a male or by plucking the secondaries of two females, suggested that secondaries play a minor role in lift force generation during hovering whereas the tip area of primaries is crucial. For the five birds, ranges of whole-bird oxygen consumption

rates, wingbeat kinematics (stroke amplitude) and lift coefficients did not vary during the moult. This constancy was mainly achieved through weight loss that alleviated aerodynamic force requirements for weight support during hovering. Since the metabolic power expenditure during moult was similar to that of normal birds but the mechanical power requirement was reduced, the flight efficiency also showed a sharp reduction during moult. This increased cost of flight may result from disruption of the integrity of the flight machinery. Overall, the control of body mass in hummingbirds can provide similar aerodynamic, muscle mechanical and physiological capacities under conditions of variable flight demand.

Key words: air density, body mass, flight feathers, heliox, hovering flight, hummingbird, *Archilochus colubris*, lift coefficient, moult, muscle power and efficiency, oxygen consumption, wing area loss.

Introduction

Moulting of the feathers must be a very energetically stressful period in the annual cycle of birds. The costs associated with moulting are manifold given that flight performance, behavioural repertoire, plumage insulation and water repellency may all be impaired (Jenni and Winkler, 1994). Furthermore, the efficiency of feather production is very low, and moulting is energetically expensive and involves processes other than just the synthesis of keratin (Lindström *et al.* 1993). On average, energy expenditures required for moulting during the moulting period are 20–40% of the basal metabolic rate (reviewed in Walsberg, 1983).

Moulting may be particularly challenging to hummingbirds because flight is essentially the only means of locomotion in this taxon, and missing flight feathers will obviously impair the mechanism of lift force generation by flapping wings. How do hovering hummingbirds compensate biomechanically? How are the aerodynamic, mechanical and physiological parameters of flight performance altered throughout a moult? This study evaluates hovering performance of ruby-throated hummingbirds during the moulting of primary flight feathers

and is the first such study investigating moulting effects on the flight energetics of birds.

Air density is a major determinant of aerodynamic power requirements (Norberg, 1990), and flight mechanics of birds can be conveniently varied by manipulation of the density of normoxic gas mixtures (Dudley and Chai, 1996). Chai and Dudley (1995) determined the limits of flight performance in hummingbirds hovering in normoxic but hypodense mixtures of air and heliox. Ruby-throated hummingbirds demonstrated considerable power reserves, and the maximum aerobic capacity at low air densities was unequivocally indicated when aerodynamic failure caused the birds to descend to the chamber floor after briefly hover-feeding. In this study, density manipulation is used to elicit variable flight energetics of moulting ruby-throated hummingbirds and to identify the limits to hovering performance.

Materials and methods

Ruby-throated hummingbirds (*Archilochus colubris* L.)

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were mist-netted in the vicinity of Austin, Texas, during the autumn migratory season of 1995 and were housed in screen cages of dimensions equal to those of the experimental cube. Hummingbird care was in accordance with federal and state guidelines. Two or three birds were housed together within one cage. The experimental procedures and equipment have been described previously (Chai and Dudley, 1995, 1996), and only a brief account will be given here. Birds were trained to feed through a cylindrical mask attached to a hanging syringe. Five individual hummingbirds with initially undamaged flight feathers (one adult male bird 1M, one juvenile male bird 2M and three female birds 3F, 4F and 5F) were used in experiments. Each bird was subjected to three trials. The first served as the control and was conducted in the winter (representing wintering, maintenance condition) prior to their moult in the spring. Two trials were conducted during the moulting of primaries (Fig. 1). Since the most stressful point during moult was probably when the bird lost most of its primaries to yield the smallest wing area and presumably smallest lift-generating capacity, one trial was timed at the point when the bird was judged to possess the smallest area of

primaries. This point occurred when the birds only had three or four old, outer primaries (the birds generally shed the first six or seven primaries over a short period). The other trial was arbitrarily chosen and was implemented either before or after the trial with smallest wing area. In this case, the old outer primaries varied in number from two to four.

Flight experiments were implemented within an airtight acrylic cube (90 cm × 90 cm × 90 cm). For each trial, data were collected initially from birds hover-feeding in unmanipulated air at pressures close to sea-level pressure. Air within the cube was then gradually replaced by filling with normoxic heliox (79% He and 21% O₂, density 0.40 kg m⁻³; the normobaric atmosphere exhibiting a density of 1.20 kg m⁻³) while allowing the cube contents to escape from an additional port. Hover-feeding flight was then recorded approximately every 15–20 min when the bird was hungry and came to feed. The duration of hover-feeding flight t was timed from video recordings. Heliox filling was terminated after the bird demonstrated aerodynamic failure in hovering; reverse pumping of ambient air was then initiated. The trial was generally stopped when the bird could then hover for increased durations and feed normally.

Density reduction associated with replacement of normal air by heliox was monitored acoustically (Dudley, 1995). Metabolic power input during hovering was obtained from measurements of rates of oxygen consumption (\dot{V}_{O_2}) using an open-flow feeder-mask respirometry system. Oxygen consumption was expressed at STP. Horizontal projections of wingbeat kinematics of each hover-feeding sequence were video-recorded from a mirror oriented above the bird at 45° to the horizontal (filming and subsequent analysis were at 60 fields s⁻¹ with a high-speed shutter of 1/4000 s). Wingbeat kinematics recorded at each hover-feeding sequence, together with morphological parameters for individual birds, were used to estimate the mechanical power requirements of flight using a detailed aerodynamic model of insect hovering flight (Ellington, 1984a–f). Wingbeat kinematics measured for each hovering sequence included wingbeat frequency n and stroke amplitude Φ . Wingbeat frequency was derived from the interaction frequency between wing motion and filming rate, and stroke amplitude was derived from video images in which the wings were located at the extreme position of the wingbeat. Morphological parameters used in aerodynamic calculations included body mass m , relative wing mass m_w for both wing pairs and expressed as a fraction of body mass, wing length R , total wing area S (the area of both wing pairs, body area not included), wing loading p_w (mg/S , where g is gravitational acceleration) and aspect ratio \mathcal{R} ($4R^2/S$). For humane reasons, none of the experimental birds was killed for *post mortem* analysis. Wing mass and its spanwise distribution were estimated from one bird of each sex in previous studies with the same species (Chai and Dudley, 1995, 1996). Wing mass and its distribution for moulting hummingbirds differ in an unknown way from those of unimpaired wings. Possible biomechanical implications of this difference for inertial power expenditure are treated in the Discussion.

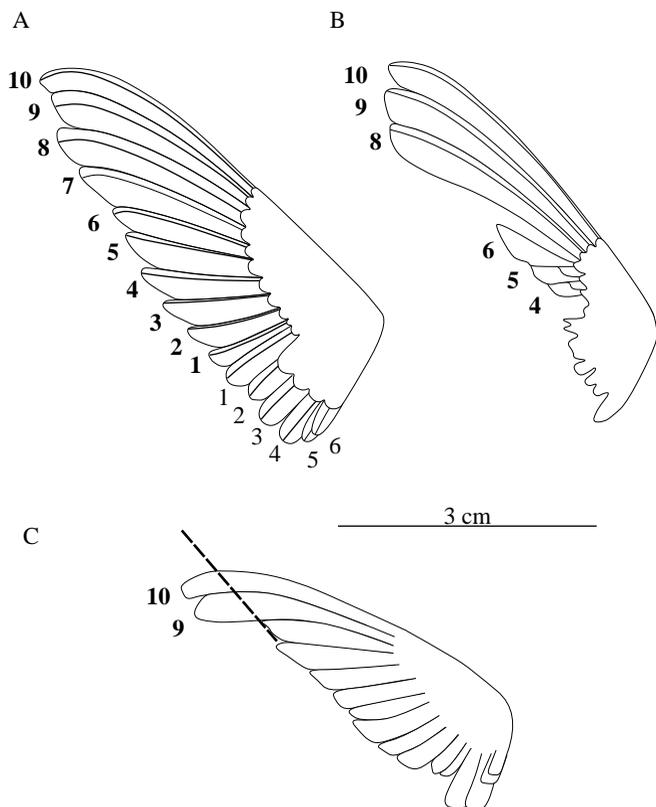


Fig. 1. Configuration of the flight feathers of moulting ruby-throated hummingbirds. (A) Unimpaired wing (bird 4F) showing primary and secondary flight feathers; moult of primaries (outward) precedes that of secondaries (inward). (B) Only three old, outer primaries 8–10 left, new primaries 4–6 growing, and all old secondaries plucked (bird 4F). (C) More advanced moult stage with two old, outer primaries 9–10 left, primaries 1–8 and secondaries 1–2 growing; the dashed line indicates where the tips were cut (bird 2M).

For each hovering flight sequence, the mechanical power requirements of flight were estimated by evaluating individual components of profile (P_{pro}^*), induced (P_{ind}^*) and inertial power during the first half of a half-stroke (P_{acc}^*). Total power expenditure for a flight sequence was calculated for the two cases of zero (P_{zero}^*) and perfect (P_{per}^*) elastic storage of wing inertial energy, which represent the maximum and minimum estimates of required mechanical power, respectively (Ellington, 1984f). Thus, $P_{\text{zero}}^* = (P_{\text{pro}}^* + P_{\text{ind}}^* + P_{\text{acc}}^*)/2$, assuming zero elastic energy storage, and $P_{\text{per}}^* = P_{\text{pro}}^* + P_{\text{ind}}^*$, assuming perfect elastic energy storage. P_{zero}^* and P_{per}^* are expressed in muscle mass-specific form, assuming that flight muscle equals 25% of the body mass (Chai and Dudley, 1995). However, during moult, flight muscle mass may not remain a constant fraction of body mass, and absolute muscle mass may not change. To evaluate this alternative assumption, a new value, $P_{\text{per}}^*(\text{fixed})$, was calculated assuming that muscle mass was fixed and did not change during moult, and that muscle mass was 25% of body mass before moult (see Table 1). Muscle mechanical efficiency η_m was estimated as $P_{\text{per}}^*/(0.9P_{\text{IN}}^*)$, assuming a 90% direct energy expenditure by flight muscle (Lasiewski, 1963; Wells, 1993). P_{IN}^* was derived from measured rates of oxygen consumption and was expressed in mass-specific form. P_{per}^* was used for comparative purposes because hummingbirds can probably store kinetic energy elastically during the deceleration phase of the wing stroke (Wells, 1993). In addition, because of body mass variation across the study period, whole-bird metabolic power input P_{IN} (body mass $\times \dot{V}_{\text{O}_2}$, unit changed to W) and whole-bird mechanical power output P_{OUT} (body mass $\times P_{\text{per}}^*$ in W) were also calculated for comparison. Thus, η_m was also equal to $P_{\text{OUT}}/(0.9P_{\text{IN}})$.

The effects of normoxic density reduction and wing area loss due to moult were evaluated for each of the ten kinematic, aerodynamic, mechanical and metabolic parameters (see Tables 1, 2) using repeated-measures analysis of variance (ANOVA) (SAS Institute, 1989). Both treatments (moult and density reduction) had three levels (see below), and these and their potential interactions were modelled as within-subject sources of variation. Because 30 tests (3×10) were conducted, sequential Bonferroni adjustment for multiple statistical tests was conducted to achieve a table-wide type I error rate of 0.05 (Holm, 1979).

Hover-feeding events were first grouped by rounding to the nearest 0.1 kg m^{-3} the air densities at which feeding events occurred. Flight performance recorded in normal air (the highest density level at 1.2 kg m^{-3}) and at aerodynamic failure (the lowest) were treated as separate density levels. Because the birds showed aerodynamic failure at different density levels ranging from 0.5 to 0.9 kg m^{-3} , a mean value across density levels between normal air and aerodynamic failure was calculated to achieve a balanced statistical design. Thus, three density levels were used for each trial. Because oxygen consumption rates at failure could not be reliably obtained given the short duration of hover-feeding, measurements at the density level closest to aerodynamic failure were used to

represent the values of the metabolic variables \dot{V}_{O_2} and P_{IN} near failure.

In addition to studying the consequences of the normal moult, an exploratory study was conducted by artificially reducing the wing area and altering the wing shape. In order to evaluate the role of primaries and secondaries in lift generation, the tips of the outer primaries (9–10) of the juvenile male (bird 2M) were cut, and all the secondaries of two females (birds 4F and 5F) were plucked (see Fig. 1, at this time, moult of secondaries had not started). These manipulations were performed after data for normal moulting had been collected.

Results

The moult pattern of captive ruby-throated hummingbirds follows the patterns described for wild individuals of this species when they moult at their wintering ground (Baltosser, 1995). The present study only concerns the moult of primaries that precedes moulting of secondaries (Fig. 1). Beginning with primary 1, moult proceeds outwards, with primaries being replaced sequentially up to primary 8, after which primary 10 is shed before primary 9. Our captive birds tended to shed primaries 1–6 or 1–7 over a relatively short period, and mainly relied on primaries 7–10 or 8–10 for flight while new primaries grew. The nonsequential moult of the outer primaries in hummingbirds is unique among birds and suggests their particular importance in flight (Epting, 1980). Individual variation with respect to the timing and rate of moult and the extent of wing area loss was also evident. Moulting was most impressive in bird 4F in that she quickly shed primaries 1–7 (loss of 30% of wing area) and flew using mainly the three outer primaries (Fig. 1B).

Density reduction always led to aerodynamic failure (Fig. 2A). The reduced wing area in moulting birds should cause failure at higher air densities than those of control birds with unimpaired wings. However, the two males actually failed at lower air densities during moult, whereas the three females, as expected, exhibited failure at higher densities. This discrepancy reflects sexual dimorphism and the differences in patterns of wing loading between sexes; male ruby-throated hummingbirds possess shorter wings and higher wing loading (Fig. 2C). When moulting flight feathers, all individuals lost body mass despite the availability of food *ad libitum* (Fig. 2B), but the extent of this loss in males was such that their wing loading was actually lower than that prior to moult, whereas wing loading remained similar before and after moulting for females. This result may explain the unexpected pattern of aerodynamic failure in the two males with respect to density reduction. In the moulting stages considered in the present study, the outer primaries (9–10, Fig. 1) were still intact, and thus the wing length was unaffected. Aspect ratio then increased with decreasing wing area (Fig. 2D).

During moult, wingbeat frequency was slightly but significantly reduced (Fig. 3; Tables 1, 2). Stroke amplitude did not change significantly, with aerodynamic failure occurring at amplitudes of 170 – 180° (Fig. 3; Tables 1, 2).

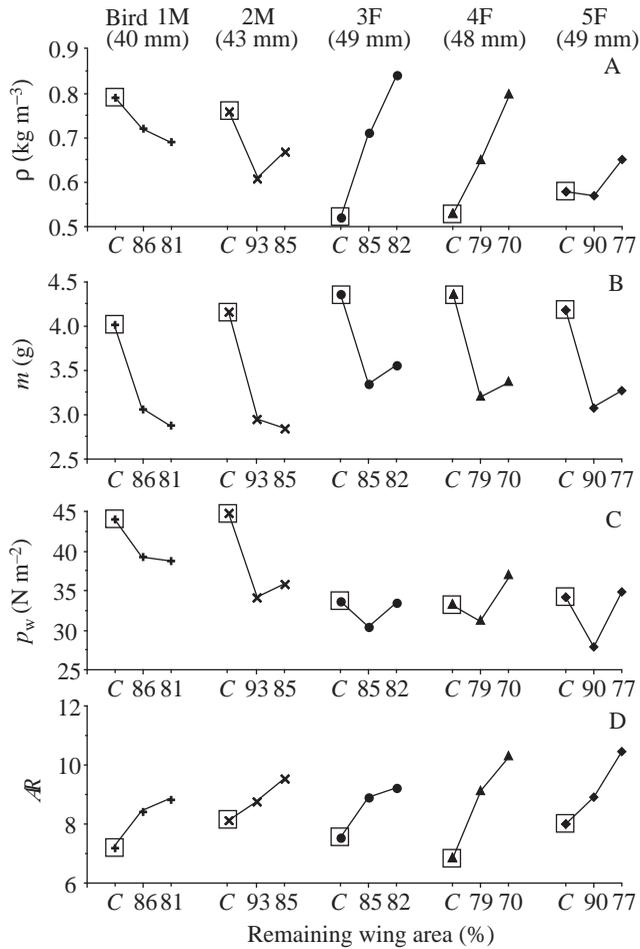


Fig. 2. Air density ρ at aerodynamic failure (A) of five individual ruby-throated hummingbirds and their morphological indicators (B, body mass m ; C, wing loading p_w ; and D, aspect ratio AR) in relation to wing area loss due to moult. Values in the boxes are from normal, unimpaired wings and serve as the control (symbol C on the x-axis). Wing length in mm is indicated for each bird.

Thus, overall wing kinematics in terms of muscle contractile velocity (proportional to wingbeat frequency \times stroke amplitude) remained similar across moulting stages. During moult, the two aerodynamic variables (mean Reynolds number \overline{Re} and mean lift coefficient $\overline{C_L}$) did not show statistically significant changes (Fig. 4; Tables 1, 2). Wing area and mean wing chord were reduced during moult, and this resulted in smaller values of \overline{Re} . However, because females showed higher failure densities at aerodynamic failure, this increase was a consequence of the decrease in wing area and resulted in an inconsistent distributional pattern of \overline{Re} between sexes. The individual P value of moult effect on \overline{Re} is 0.006 (Table 2). However, after sequential Bonferroni adjustment, this P value became marginally insignificant. The distributional pattern of $\overline{C_L}$ did not indicate any effect of moult. Depending on the extent of elastic energy storage, the effect of moult on muscle mass-specific power output could be significant (Fig. 5; Tables 1, 2). While P_{zero}^* showed no trend

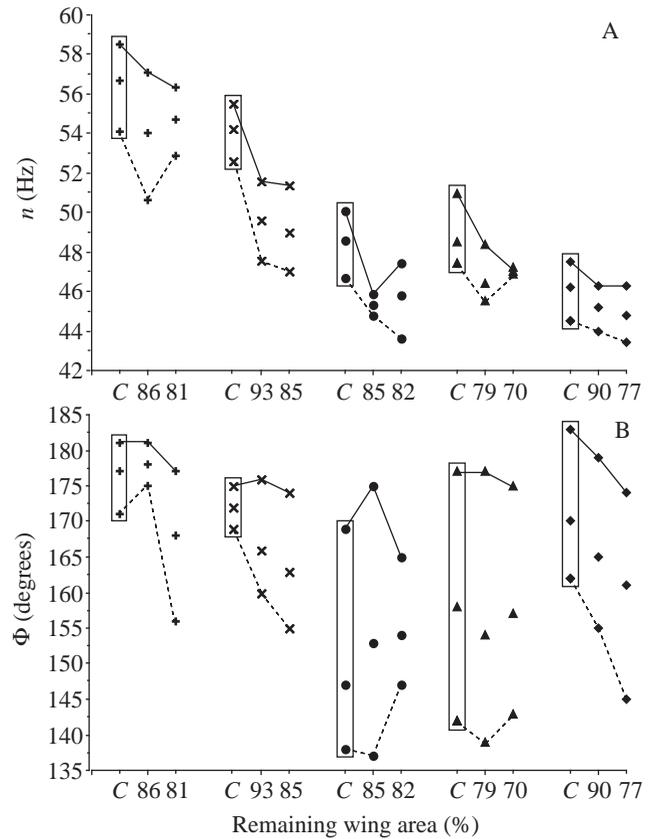


Fig. 3. Wingbeat frequency n (A) and stroke amplitude Φ (B) in relation to wing area loss due to moult and air density reduction with heliox (order of individuals as in Fig. 2). Solid lines link values at aerodynamic failure; dashed lines link values in normal air. Intermediate values are means across density levels. Values in the boxes are from normal, unimpaired wings and serve as the control (symbol C on the x-axis).

(Fig. 5A), P_{per}^* with reduced wing area showed a generally downward trend (Fig. 5B). \dot{V}_{O_2} showed a steep and significant increase (Fig. 6A) with an associated and significant decrease in muscle efficiency η_m (Fig. 6B; Tables 1, 2). During moult, all birds showed reduced body masses. As a result, patterns of whole-bird metabolic and mechanical performance were substantially different from their mass-specific values (Fig. 7; Tables 1, 2). P_{IN} showed no significant upward trend, whereas P_{OUT} showed a steep decline with reduced wing area. Statistical results for density effects on kinematic, mechanical and metabolic variables were in close agreement with earlier studies on normoxic density reduction using heliox (Chai and Dudley, 1995, 1996). Density reduction significantly affected all variables except for muscle mechanical efficiency; η_m was, however, significantly influenced by moult (Table 2).

Artificial wing area and shape manipulation

In the exploratory study (no statistics were performed owing to the small number of samples), cutting off the tips of primaries 9 and 10 of bird 2M resulted in a smaller loss of wing area (10%) than plucking out all the secondaries (17% for bird

Table 1. Summary of morphological, kinematic, mechanical and metabolic variables for five birds in relation to moulting stages

Variables	Moulting stages		
	Before moult, wings unimpaired	Moulting, intermediate wing area loss	Moulting, greatest wing area loss
ρ_{fail} (kg m ⁻³)	0.64±0.13	0.65±0.06	0.73±0.08
m (g)	4.2±0.1	3.1±0.2	3.2±0.3
p_w (N m ⁻²)	38.0±5.9	32.6±4.4	36.0±2.0
\mathcal{R}	7.5±0.5	8.8±0.3	9.7±0.7
n (Hz)	49±4	47±3	47±4
	53±4	50±5	50±4
Φ (degrees)	156±15	153±16	149±6
	177±5	178±2	173±5
\overline{Re}	9100±500	7200±400	6500±400
	5400±800	4600±500	4700±900
$\overline{C_L}$	1.38±0.12	1.38±0.17	1.49±0.11
	1.75±0.10	1.62±0.26	1.61±0.23
P_{zero}^* (W kg ⁻¹)	250±25	254±23	240±21
	343±18	373±22	350±25
P_{per}^* (W kg ⁻¹)	101±4	93±2	95±3
	131±6	119±2	117±4
P_{per}^* (fixed)	101±4	69±3	72±6
	131±6	88±2	88±6
\dot{V}_{O_2} (ml O ₂ g ⁻¹ h ⁻¹)	41±2	53±3	56±8
	47±4	64±6	67±5
η_m for P_{per}^* (%)	11.6±0.3	8.3±0.4	8.2±0.9
	12.7±1.0	8.6±1.1	8.1±0.3
P_{IN} (W)	1.02±0.05	0.98±0.07	1.04±0.16
	1.16±0.11	1.18±0.16	1.25±0.11
P_{OUT} (W)	0.106±0.004	0.073±0.005	0.075±0.008
	0.138±0.011	0.093±0.005	0.093±0.008

The first value of kinematic, mechanical and metabolic variables refers to the value in normal air; the second to the value at aerodynamic failure (\dot{V}_{O_2} and P_{IN} close to aerodynamic failure).

Values at muscle mass-specific power output, P_{zero}^* and P_{per}^* , were calculated assuming that flight muscle equals 25 % of the body mass.

P_{per}^* (fixed), recalculated P_{per}^* assuming that muscle mass is fixed and does not change during moult and that muscle mass was 25 % of body mass before moult.

All values are given as mean ± s.d., $N=5$.

ρ_{fail} , air density at aerodynamic failure; m , body mass; p_w , wing loading; \mathcal{R} , aspect ratio; n , wingbeat frequency; Φ , stroke amplitude; \overline{Re} , mean Reynolds number; $\overline{C_L}$, mean lift coefficient; P_{zero}^* and P_{per}^* , mechanical power output per unit flight muscle mass assuming zero and perfect elastic energy storage; \dot{V}_{O_2} , body mass-specific rate of oxygen consumption; η_m for P_{per}^* , muscle mechanical efficiency; P_{IN} , whole-bird metabolic power input; P_{OUT} , whole-bird mechanical power output assuming perfect elastic energy storage.

4F and 18 % for 5F; see Fig. 1 and Table 3). However, the reduction in wing length and wing tip velocity for bird 2M caused by trimming the primaries exerted more drastic aerodynamic consequences than the loss of secondaries, as

Table 2. Results of repeated-measures ANOVA for kinematic, mechanical and metabolic variables assessing effects of wing area reduction due to moult and air density reduction in heliox

Variables	P value		
	Moult	Density	Moult × Density
n	0.001* (2,8)	0.001* (2,8)	0.927 (4,16)
Φ	0.058 (2,8)	0.001* (2,8)	0.780 (4,16)
\overline{Re}	0.006 (2,8)	0.001* (2,8)	0.001* (4,16)
$\overline{C_L}$	0.746 (2,8)	0.003* (2,8)	0.021 (4,16)
P_{zero}^*	0.030 (2,8)	0.001* (2,8)	0.176 (4,16)
P_{per}^*	0.001* (2,8)	0.001* (2,8)	0.242 (4,16)
\dot{V}_{O_2}	0.001* (2,8)	0.001* (2,8)	0.146 (4,16)
η_m for P_{per}^*	0.001* (2,8)	0.136 (2,8)	0.130 (4,16)
P_{IN}	0.269 (2,8)	0.001* (2,8)	0.466 (4,16)
P_{OUT}	0.001* (2,8)	0.001* (2,8)	0.002* (4,16)

P values are from univariate F-test.

d.f. (given in parentheses), degrees of freedom.

* significant on the basis of sequential Bonferroni adjustment with table-wide type-I error rate at 0.05.

n , wingbeat frequency; Φ , stroke amplitude; \overline{Re} , mean Reynolds number; $\overline{C_L}$, mean lift coefficient; P_{zero}^* and P_{per}^* , mechanical power output per unit flight muscle mass assuming zero and perfect elastic energy storage; \dot{V}_{O_2} , body mass-specific rate of oxygen consumption; η_m for P_{per}^* , muscle mechanical efficiency; P_{IN} , whole-bird metabolic power input; P_{OUT} , whole-bird mechanical power output assuming perfect elastic energy storage.

indicated by a greatly increased air density at aerodynamic failure (Fig. 8, Table 3). The two females could tolerate total loss of secondaries with only a slight increase of air density at aerodynamic failure. Bird 4F was already operating with a much reduced wing area, only 79 % that of unimpaired wings, because of her relatively fast rate of moulting, and after plucking out all her secondaries, she was flying with 62 % of pre-moult wing area. Despite this, air density at aerodynamic failure remained similar to that occurring before feather plucking (Fig. 8), although both failure densities were higher than that measured before the moult (0.53 kg m⁻³, Fig. 2A).

For all three birds, artificial reduction of wing area shortened the hover-feeding duration (Fig. 8). This reduction was most evident for hovering duration in normal air, which tended to be the longest in duration. Air density reduction using heliox also shortened hover-feeding duration and eventually led to aerodynamic failure. Hover-feeding duration at aerodynamic failure was very short and probably represented transient burst performance (Chai and Dudley, 1996), whereas hover-feeding in normal air represented sustainable flight. Thus, the duration of sustained flight of all three birds was adversely affected by artificial wing area reduction, whereas only the transient performance of bird 2M was greatly curtailed. After cutting the tips of the two outer primaries of bird 2M, the steep increase in mean lift coefficient at aerodynamic failure (to a maximum value of 2.35, Table 3) indicated that its wings had difficulty generating sufficient lift force. This was probably the reason

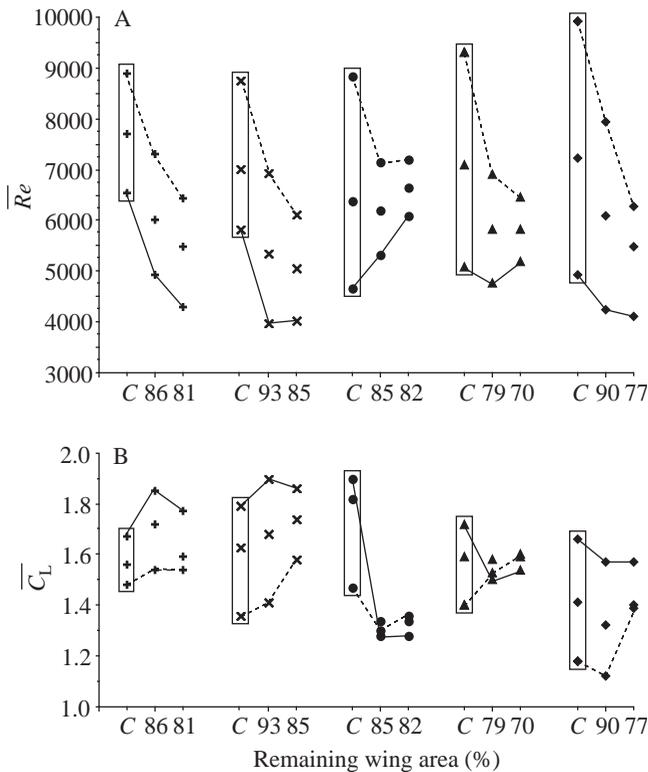


Fig. 4. Mean Reynolds number \overline{Re} (A) and mean lift coefficient \overline{Cl} (B) in relation to moulting and air density reduction (order of individuals as in Fig. 2). Solid lines link values at aerodynamic failure; dashed lines link values in normal air. Intermediate values are means across density levels. Values in the boxes are from normal, unimpaired wings and serve as the control (symbol C on the x-axis).

for aerodynamic failure at a much higher air density. However, this bird compensated for its wing damage by changing its wingbeat kinematics after 8 days (both wingbeat frequency and stroke amplitude increased) to improve lift-generating capacity, giving lift coefficients comparable to those before the operation (Table 3). (During this period, bird 2M had shed primary 10 but primary 8 grew longer, so that the total wing area remained the same.) Hover-feeding duration in normal air was lengthened, but air density at aerodynamic failure remained similar to that immediately following the operation (Fig. 8, cf. After 1 and 2).

Discussion

Wing mass and its distribution in moulting hummingbirds were not measured in the present study, and values from birds with unimpaired wings were used instead. Since no noticeable change in muscle distribution along the wing was evident (P. Chai, personal observation) and hummingbird wing mass is primarily restricted to the wing base (Wells, 1990), wing mass and its distribution in moulting hummingbirds were unlikely to show significant variation. Both wing area and moment of inertia for the virtual mass of the wing pair were reduced in moulting hummingbirds, and this should be correlated with

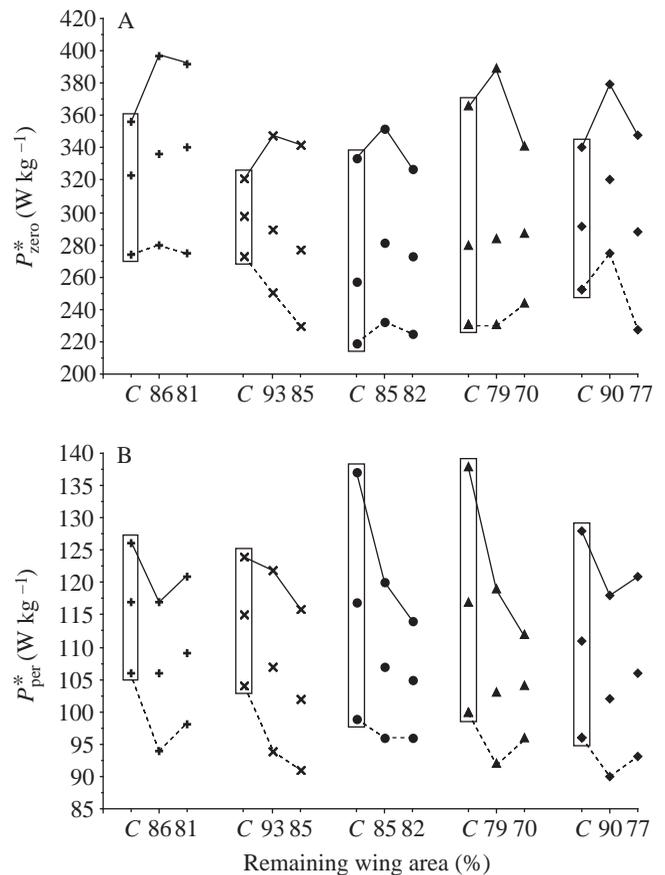


Fig. 5. Muscle mass-specific power output assuming zero (P_{zero}^* , A) and perfect (P_{per}^* , B) elastic energy storage in relation to moulting and air density reduction (order of individuals as in Fig. 2). P_{zero}^* and P_{per}^* were calculated assuming that flight muscle equals 25% of the body mass. Solid lines link values at aerodynamic failure; dashed lines link values in normal air. Intermediate values are means across density levels. Values in the boxes are from normal, unimpaired wings and serve as the control (symbol C on the x-axis).

increased wingbeat frequency (Pennycuik, 1990). However, wingbeat frequency was slightly but significantly reduced (Fig. 3). This reduction is difficult to explain, but may relate to physiological changes in the thoracic musculature or increased variability in wingbeat kinematics associated with feather loss.

Various physiological systems involved in neural control, oxygen and fuel supply, muscle mechanics, wingbeat kinematics, etc., must be efficiently linked for a bird to function in flight. Each such system should be flexible enough to respond to variable ecological and physiological demands on flight, such as those occurring during migration, courtship and moulting (Rayner, 1988, 1995). In the present study, it has been shown that ruby-throated hummingbirds maintained their ability to hover given a wide range of body masses, air densities, wing area and wing shape and also that muscle mass-specific rates of oxygen consumption could vary substantially. This finding suggests broad responsiveness in the flight motor. However, the ranges of whole-bird oxygen consumption rate, wingbeat kinematics and lift coefficients remained similar,

Table 3. Artificial wing area reduction on three ruby-throated hummingbirds during moult and the effects on morphological, kinematic, mechanical and metabolic variables

Variables	Tips of two outer primaries cut			All secondaries plucked			
	Bird 2M			Bird 4F		Bird 5F	
	Before	After 1	After 2	Before	After	Before	After
Wing area reduction (%)	5	15	15	21	38	10	28
ρ_{fail} (kg m^{-3})	0.71	0.87	0.88	0.65	0.66	0.57	0.60
m (g)	2.84	2.79	2.80	3.21	3.14	3.07	3.16
R (mm)	43	36	36	48	48	49	49
p_w (N m^{-2})	31.9	35.0	35.5	31.2	38.7	27.9	35.9
\mathcal{A}	8.5	6.6	6.7	9.1	11.6	8.9	11.1
n (Hz)	48	50	53	46	45	44	45
Φ (degrees)	51	51	56	48	49	46	48
	153	169	172	139	141	155	148
\overline{Re}	177	175	181	177	175	179	172
	6900	7000	7600	6900	5500	7900	6200
$\overline{C_L}$	4900	5400	6200	4800	3800	4200	3600
	1.52	1.98	1.49	1.53	1.44	1.12	1.40
P_{zero}^* (W kg^{-1})	1.64	2.35	1.62	1.50	1.44	1.57	1.74
	243	234	290	231	239	275	252
P_{per}^* (W kg^{-1})	356	258	355	389	403	379	381
	90	96	100	92	96	90	90
\dot{V}_{O_2} ($\text{ml O}_2 \text{g}^{-1} \text{h}^{-1}$)	111	109	114	119	126	118	118
	51	57	60	57	59	50	49
η_m for P_{per}^* (%)	66	59	68	69	71	67	68
	8.3	8.0	7.9	7.8	7.8	8.5	8.7
P_{IN} (W)	7.8	8.1	7.7	7.2	7.5	8.1	7.8
	0.85	0.94	0.98	1.07	1.08	0.90	0.91
P_{OUT} (W)	1.10	0.97	1.12	1.29	1.30	1.20	1.25
	0.064	0.067	0.070	0.074	0.075	0.069	0.071
	0.079	0.076	0.080	0.096	0.099	0.090	0.093

Because of moult, wing area was already reduced relative to unimpaired wings (under Before) and was further reduced after the operation (under After).

Kinematic, mechanical and metabolic variables show two values taken (first) in normal air and (second) at aerodynamic failure (\dot{V}_{O_2} and P_{IN} taken close to aerodynamic failure).

For each bird, all measurements were collected on the same day except for values under After 2 (bird 2M), which were collected 8 days later.

Wing area reduction indicates the percentage reduction relative to unimpaired wings before moult; ρ_{fail} , air density at aerodynamic failure; m , body mass; R , wing length; p_w , wing loading; \mathcal{A} , aspect ratio; n , wingbeat frequency; Φ , stroke amplitude; \overline{Re} , mean Reynolds number; $\overline{C_L}$, mean lift coefficient; P_{zero}^* and P_{per}^* , mechanical power output per unit flight muscle mass assuming zero and perfect elastic energy storage; \dot{V}_{O_2} , body mass-specific rate of oxygen consumption; η_m for P_{per}^* , muscle mechanical efficiency; P_{IN} , whole-bird metabolic power input; P_{OUT} , whole-bird mechanical power output assuming perfect elastic energy storage.

pointing to homeostasis and a set range of values to which an individual hummingbird is adapted. Mass loss can alleviate aerodynamic force requirements for weight support, and, therefore, by controlling body mass in relation to different flight demands, hummingbirds can maintain a sufficient aerodynamic performance. At the same time, physiological parameters in relation to muscle contractile dynamics and oxygen and fuel supply can be maintained without excessive variation. However, mass loss presumably reduces fat reserves and may jeopardize energy security. Thus, such a strategy may be selected for only under extraordinary flight conditions.

More studies are obviously needed to evaluate the role of body mass control in hummingbirds during moult and other flight activities such as migration (Carpenter *et al.* 1993) and courtship (Calder *et al.* 1990). Two other studies also showed that moulting hummingbirds in captivity with reduced wing area typically lost mass (Wells, 1990; Hiebert, 1993), and this also appears to be the case for other birds (e.g. King, 1968; Klaassen, 1995).

Dissimilar body masses prior to and during moult point to differences in body composition. In the present study, P_{per}^* and P_{zero}^* were calculated assuming that flight muscle equals 25 %

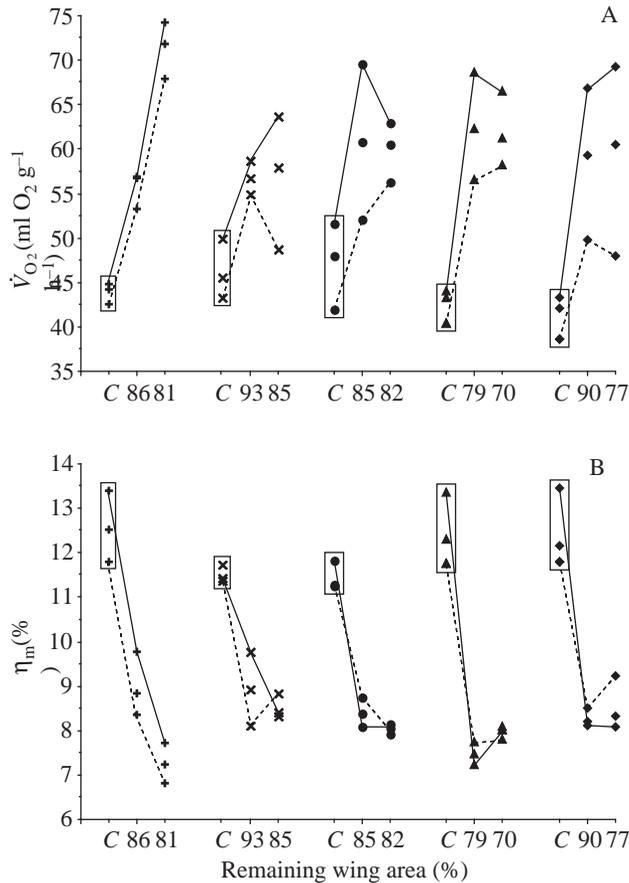


Fig. 6. Body mass-specific rate of oxygen consumption \dot{V}_{O_2} (A) and muscle mechanical efficiency η_m (B) assuming perfect elastic energy storage in relation to moult and air density reduction (order of individuals as in Fig. 2). Solid lines link values close to aerodynamic failure; dashed lines link values in normal air. Intermediate values are means across density levels. Values in the boxes are from normal, unimpaired wings and serve as the control (symbol C on the x-axis).

of body mass. However, it is not possible to determine accurately the proportions of flight muscle and fat storage in relation to varying body mass without immediate *post mortem* analysis (ruby-throated hummingbirds can undergo rapid body mass change of up to 10% in 1 day; P. Chai, unpublished material). The values of P_{per}^* and $P_{per}^*(fixed)$ represent the two extremes (Table 1) in that P_{per}^* assumes that flight muscle mass remains a constant fraction of body mass whereas $P_{per}^*(fixed)$ assumes that flight muscle mass is invariant before and during moult. The patterns of muscle and fat changes during moult are presently unclear.

The model developed by Ellington (1984f) uses approximations to calculate profile drag coefficient ($C_{D,pro}$); thus, the value of P_{pro}^* is less accurate than the value of P_{ind}^* (P_{pro}^* generally only accounts for one-quarter of P_{per}^*). Profile drag and P_{pro}^* may be higher for a moulting bird with imperfect wings. The wing shape of moulting hummingbirds also deviates from Ellington's (1984b) 'laws of wing shape', i.e. strong positive correlations between 'shape parameters' of spanwise wing area, mass and virtual mass distributions for a

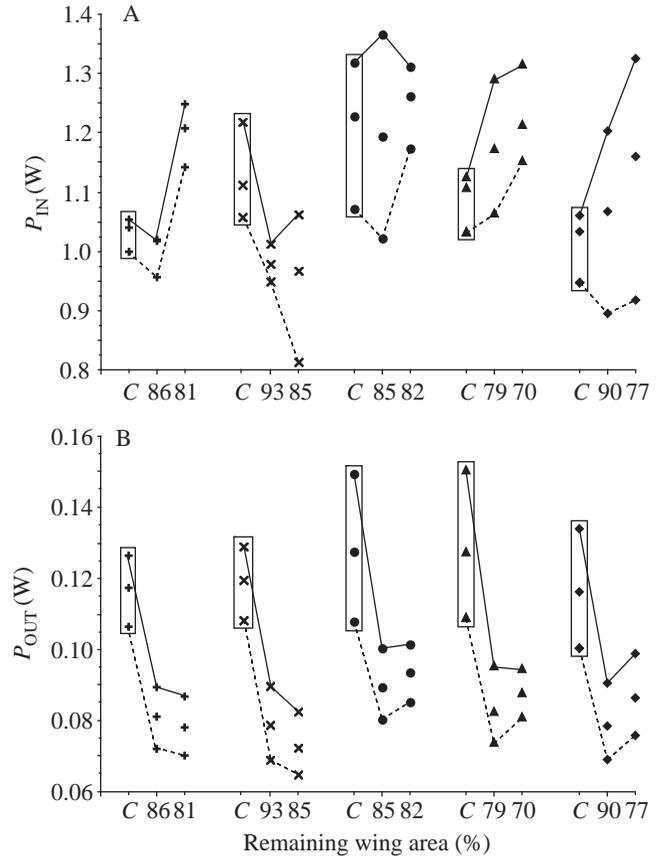


Fig. 7. Whole-body metabolic power input P_{IN} and whole-body mechanical power output P_{OUT} in relation to moult and air density reduction (order of individuals as in Fig. 2). Solid lines link values at or close to aerodynamic failure; dashed lines link values in normal air. Intermediate values are means across density levels. Values in the boxes are from normal, unimpaired wings and serve as the control (symbol C on the x-axis).

diverse collection of insects, birds and bats. An explanation for these tight relationships is still elusive, but the reduced muscle mechanical efficiency (η_m) of moulting birds may indicate a 'penalty' for straying from the laws of shape. To understand the mechanisms and dynamics of aerodynamic force generation of flapping wings more fully, further detailed investigation, using techniques such as high-speed video and vortex visualization to measure angle of attack, wing twist, and vortex formation and shedding, is needed.

Both whole-bird and mass-specific power output (P_{OUT} and P_{per}^*) declined during moult, whereas whole-bird power input (P_{IN}) remained stable and mass-specific rate of oxygen consumption (\dot{V}_{O_2}) increased. Thus, hovering during moult became more costly and muscle efficiency η_m was significantly reduced. To facilitate comparison, η_m was calculated assuming that 10% of metabolic power was used by basal metabolism and maintenance needs, i.e. 90% direct energy expenditure by flight muscles (Lasiewski, 1963; Wells, 1993). However, energy expenditures dedicated to moulting average 20–40% of the basal metabolic rate (Walsberg, 1983). Even adjusting for this bias, the muscle mechanical efficiency

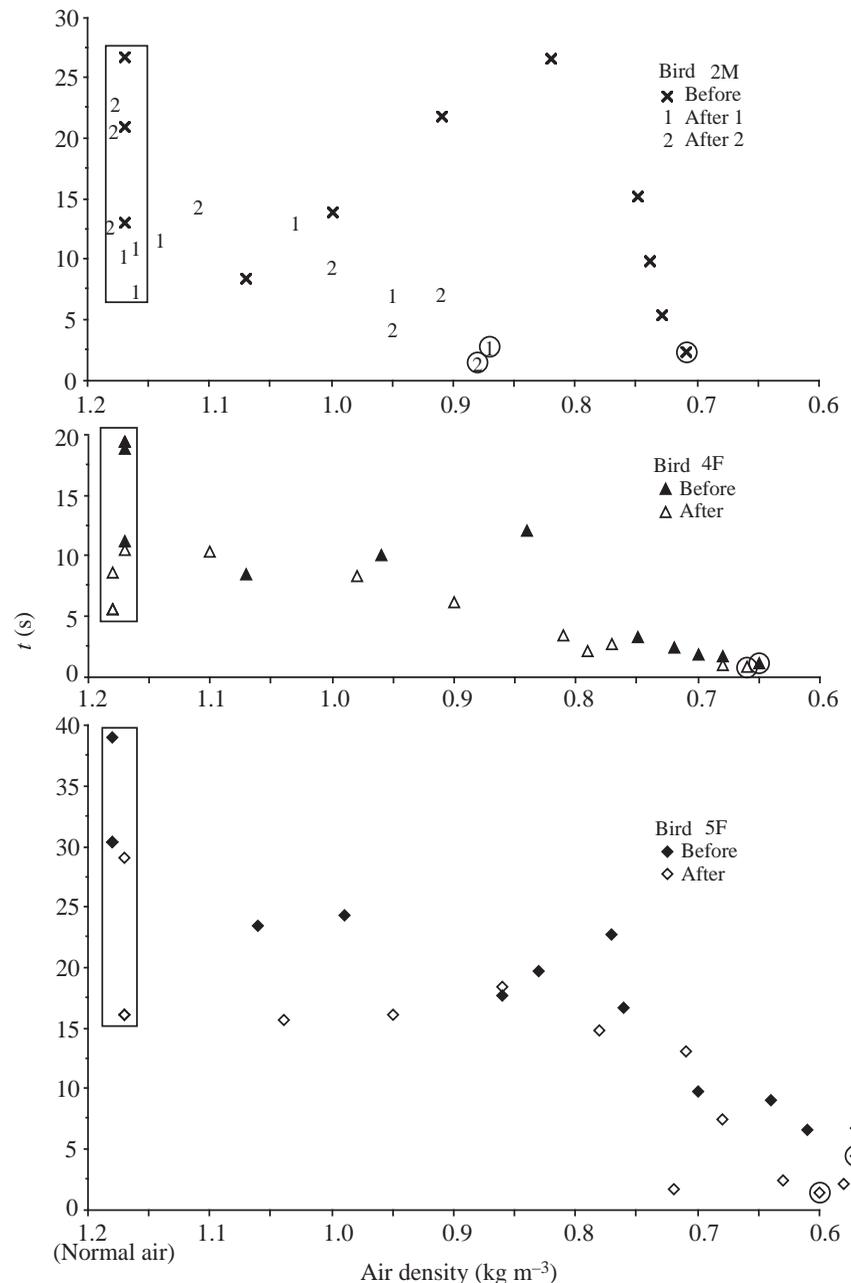


Fig. 8. Hover-feeding duration t before and after artificial wing area reduction and in relation to air density reduction. The tips of the two outer primary flight feathers of bird 2M were cut, and all secondaries of the two females (birds 4F and 5F) were plucked. Values in the boxes are from normal air; circled values are at aerodynamic failure. For each bird, all measurements (before and after the operation; after 1 for bird 2M) were from the same day except for values of After 2 (bird 2M), which were collected 8 days later.

during moult was still greatly reduced (Fig. 6). Hummingbirds with damaged wings also augment their oxygen consumption and exhibit reduced flight efficiency (Epting, 1980; Wells, 1990).

Although ruby-throated hummingbirds demonstrate flexible physiological functions to cope with different flight demands, the greatly reduced η_m during moult suggests that there is an ideal flight design which links the aerodynamic properties of the wings, the contractile properties of the flight muscles and the physiological features of oxygen and fuel supply. Moult disrupts the integrity of this flight machinery. For example, for a hummingbird to hover efficiently, wing morphology and wingbeat kinematics should presumably operate in such a way that the supporting flight muscles can perform at their optimal

strain and rates of strain with minimal risk of damaging bones, muscles and tendons. Concurrently, the force balance achieved by the flapping wings should be stable. Impaired wings may increase the operating cost of flight muscles and their regulation of flight stability through neural control.

Artificial manipulation of flight feathers suggests that secondaries play a minor role in lift force generation during hovering, whereas the tip area of primaries is crucial. This result is consistent with the hovering aerodynamic model of Ellington (1984*f*). After cutting the tips of the two outer primaries of bird 2M, mean lift coefficients rose sharply (see After 1, Table 3). Such excessive lift requirements probably contributed to early aerodynamic failure for this bird. The \dot{V}_{O_2} of bird 2M close to aerodynamic failure at After 1 was lower

than that before the operation. This result suggests that oxygen supply was not limiting immediately after the operation. However, 8 days later, at After 2, bird 2M was able to meet force requirements by altering wingbeat kinematics. The duration of hover-feeding in normal air was restored to the level occurring before the operation, although air density at aerodynamic failure did not change (Fig. 8). Since the \dot{V}_{O_2} of bird 2M close to aerodynamic failure at After 2 was similar to the level before the operation, oxygen supply may have been a limiting factor. Although bird 2M showed similar failure densities at After 1 and 2, two different types of constraints might be responsible. Cutting the tips of primaries and plucking out secondaries obviously caused further disruption to flight integrity, which was already degraded through moulting. Toleration of damage and the restoration of performance present an interesting experimental context within which to evaluate flight capacity, and both natural and artificial disturbance of flight feathers can shed light on the relative roles of aerodynamic, mechanical and physiological constraints. The results of the exploratory manipulations discussed in this paper point to complex physical and biological factors influencing both sustainable and transient capacities to generate lift force by flapping wings.

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